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JC674 U.S. PTO

September 8, 1999

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Sir:

Transmitted herewith for filing is the
patent application of:

Inventor: Gary L. Owen et al

For: Method And Apparatus For Interactively Displaying A Route Window
For A Flight Management System

Doc. No.: 99CR108/KE

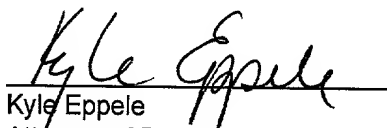
Enclosed are:

- ☒ 6 Sheet of Drawings.
- ☒ An Assignment of Invention to Rockwell Collins, Inc., Cedar Rapids, IA 52498.
- ☒ Combined Declaration and Power of Attorney.
- ☒ Information Disclosure Statement and references.

For	Number Filed	Claims as Filed Number Extra	Rate	Base Fee	
Total Claims	20 - 20	0	X \$18		\$760.00
Ind. Claims	3 - 3	0	X \$78		\$ 0.00
Total Filing Fee					\$ 760.00

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Respectfully submitted,


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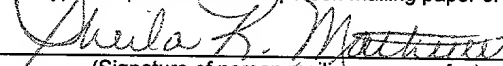
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Sheila K. Mathews

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09/391782
09/08/99

METHOD AND APPARATUS
FOR INTERACTIVELY DISPLAYING

A

ROUTE WINDOW FOR

A

FLIGHT MANAGEMENT SYSTEM

By

GARY L. OWEN,

SARAH BARBER,

AND

GEORGE W. PALMER

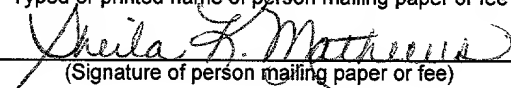
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CROSS REFERENCE TO RELATED PATENT APPLICATIONS

This application is related to the below listed co-pending patent applications which are filed on even date herewith, are assigned to the same assignee, and are incorporated herein in their entirety by these references:

An application entitled "Method and Apparatus For Interactively Selecting Display Parameters For An Avionics Flight Display" by Sarah Barber, Norm W. Arons, and George W. Palmer;

An application entitled "Method and Apparatus For Interactively Selecting, Controlling and Displaying Parameters For An Avionics Radio Tuning Unit" by George W. Palmer, Claude Eyssautier, and Matt Smith;

An application entitled "Method and Apparatus For Graphically Inserting Waypoints For A Flight Management System" by Martin Pauly; and

An application entitled "Method And Apparatus For Interactively And Automatically Selecting, Controlling And Displaying Parameters For An Avionics Electronic Flight Display System" By Matt Smith and Gary L. Owen.

FIELD OF THE INVENTION

The present invention generally relates to avionics, and more particularly relates to flight management systems (FMSs), and even more particularly relates to combined electronic displaying of textual FMS and graphical map data.

BACKGROUND OF THE INVENTION

In the past, designers of avionics displays and flight computer systems have endeavored to achieve a reduction in pilot workload. One area of concern has been the FMS, which typically requires a significant amount of "heads-down" time. This "heads-down" time occurs when the pilot is neither looking at the primary flight displays, nor out the wind screen, but instead is focused upon a task in an oblique direction, such as when using a typical FMS control display unit (CDU), which is an integrated keyboard and a textual display unit. One approach has been proposed in which a portion of the typical FMS textual information is displayed on a segment of a multi-functional display also containing a map display. A cursor is used to select textual information to be displayed on the map display segment. While these interactive dual format displays have clear advantages, they also have significant drawbacks.

The cockpit can become, at times, a very busy place. During take-off and approach, the workload on a pilot can be extreme. Since the proposed dual format displays are capable of showing only a limited amount of textual information, they still require the pilot to make reference to additional pages or displays. This additional effort, at an already busy time, is quite undesirable.

Consequently, there exists a need for improved methods and apparatuses for displaying both textual FMS data, as well as graphical FMS maps.

SUMMARY OF THE INVENTION

It is an object of the present invention to provide an improved FMS display.

It is a feature of the present invention to utilize a route window which simultaneously displays all textual information relating to a flight plan leg that is displayed on a graphical FMS map display.

It is an advantage of the present invention to reduce heads-down time and thereby enhance situational awareness by providing a single source for all textual information required for a flight plan leg.

It is another feature of the present invention to include a configurable route window size characteristic.

It is another advantage of the present invention to provide for increased control of the amount of textual information displayed at one time.

It is yet another feature of the present invention to include a configurable route window content characteristic.

It is yet another advantage of the present invention to provide for the capability of reducing pilot workloads at critical times, by permitting the route window to be configured in response to current events, such as weather, air traffic, phase of flight (take-off, approach, etc.) to show the most important and timely types of information required by the pilot.

The present invention is an apparatus and method for simultaneously displaying FMS textual data and FMS map data, which is designed to satisfy the aforementioned needs, provide the previously stated objects, include the above-listed features, and achieve the already articulated advantages. The present invention is carried out in a "fixed format-less" manner in a sense that the undesirable constraints of a fixed format textual display segment have been greatly reduced.

Accordingly, the present invention is a multi-mode interactive FMS display which simultaneously displays FMS map data, as well as displays FMS textual data in a configurable route window.

BRIEF DESCRIPTION OF THE DRAWINGS

The invention may be more fully understood by reading the following description of the preferred embodiments of the invention, in conjunction with the appended drawing wherein:

Figure 1 is a simplified representation of a prior art display, showing both textual and map segments.

Figure 2 is a representation of a display of the present invention, wherein the box in the upper left-hand corner is a route window.

Figure 3 is a representation of a display of the present invention wherein the box in the upper portion is a horizontally expanded route window of Figure 2.

Figure 4 is a representation of a display of the present invention wherein the box in the upper portion includes alternate parameters with respect to those shown in the route window of Figure 3.

Figure 5 is a representation of a display of the present invention wherein the box in the left portion is a vertically expanded route window of Figure 2.

Figure 6 is a representation of a display of the present invention wherein the entire display is a horizontally and vertically expanded route window of Figure 2.

DETAILED DESCRIPTION

Now referring to the drawings wherein like numerals refer to like matter throughout, there is shown in Figure 1, a display of the prior art, generally designated 100, having a lower section 102, which is a graphical FMS map in a plan position indication format, showing a flight plan leg segment with a centrally disposed leg having an endpoint 106 labeled "CID." Also shown in a top portion of display 100 is FMS textual display segment 104, which shows a limited amount of the information (only waypoints). If the pilot needs additional FMS textual information, then the pilot is required to resort to a prior art CDU .

Now referring to Figure 2, there is shown an improved FMS display of the present invention, generally designated 200, showing an adaptive graphical map segment 202 and a configurable route window 204 in the upper left-hand corner. Configurable route window 202 exhibits a graphical user interface (GUI), which is responsive to a cursor device manipulated by the pilot. Adaptive graphical map segment 202 is preferably adaptive in size, position and orientation to the particular configuration of configurable route window 204; however, a static graphical map segment could conceivably be used in an alternate embodiment. For example, adaptive graphical map segment 202 shows graphical map information in all areas other than that which is consumed by configurable route window 204. Additionally, the configurable route window 204 could be displayed without any map of any type, this could be thought of as configuring the configurable route window 204 to a maximum possible size. Configurable route

window 204 includes display format parameter control buttons 206, which are variable buttons to give a pilot individual control over the content of the information to be displayed. Display format parameter control buttons 206 currently show WP and WP/RT on one button and SPD and VPA on another button. Note that WPT and SPD are highlighted to show their selection status. Vertical route window expansion/contraction variable direction double chevron button 208 is shown in the lower left-hand corner of configurable route window 204. Vertical route window expansion/contraction variable direction double chevron button 208 is used to expand or contract the vertical size of configurable route window 204. Depending upon the current status (expanded or normal), the directions of the two chevrons thereon are variable. Vertical route window expansion/contraction variable direction double chevron button 208 is shown here with a downward directed chevron, which indicates that the configurable route window 204 is capable of being expanded downward into the free space below it upon clicking vertical route window expansion/contraction variable direction double chevron button 208. A route window removal x button 210 is included which will result in a removal of the configurable route window 204 when clicked upon. Similar to vertical route window expansion/contraction variable direction double chevron button 208, a horizontal route window expansion/contraction variable direction double chevron button 212 is shown. Horizontal route window expansion/contraction variable direction double chevron button 212 is used to expand or contract the horizontal size of the configurable route window 204. Depending upon the current status (expanded or normal), the

directions of the two chevrons thereon are variable. Horizontal route window expansion/contraction variable direction double chevron button 212 is shown here with a rightward directed chevron which indicates that the configurable route window 204 is capable of being expanded rightward into the free space to the right of it upon clicking horizontal route window expansion/contraction variable direction double chevron button 212.

Now referring to Figure 3, there is shown a display of Figure 2, generally designated 300, where the horizontal route window expansion/contraction variable direction double chevron button 212 has been clicked to expand the configurable route window 204. Adaptive graphical map segment 202 has adapted to the configuration of configurable route window 204. Horizontal route window expansion/contraction variable direction double chevron button 212 now shows its variable double chevrons pointed in a leftward direction to indicate the capability of contracting the configurable route window 204 in a leftward direction upon clicking of horizontal route window expansion/contraction variable direction double chevron button 212. Vertical route window expansion/contraction variable direction double chevron button 208 is unchanged with respect to Figure 2. Now with the expanded horizontal size of configurable route window 204 display format parameter control buttons 206 have grown in number to include additional parameters of ETE and ETA on one new button and WND and ISA on another new button. Note that ETE and WND are highlighted to show their selection status. Clearly, the clicking of horizontal route window expansion/contraction

variable direction double chevron button 212 had resulted in a reconfiguration of configurable route window 204 to a larger horizontal size thereby allowing for more textual FMS data to be displayed.

Now referring to Figure 4, there is shown a display of Figure 3 where the configurable route window 204 remains the same in size, but the content of the parameters selected has changed. Display format parameter control buttons 206 have had each of their parameters changed. Note that WPT/RT, VPA, ETA and ISA are now all highlighted to show their new selection status. The information displayed below these buttons with changed highlighting has correspondingly changed in response to the new selection status of each button. The size, orientation and direction of adaptive graphical map segment 202, vertical route window expansion/contraction variable direction double chevron button 208 and horizontal route window expansion/contraction variable direction double chevron button 212 are unchanged with respect to Figure 3 because neither vertical route window expansion/contraction variable direction double chevron button 208, nor horizontal route window expansion/contraction variable direction double chevron button 212, were clicked and, therefore, adaptive graphical map segment 202 has no size and orientation changes from which it needs to respond.

Now referring to Figure 5, there is shown a display, generally designated 500, where the configurable route window 204 has been vertically expanded, in relation to Figure 2, in response to a clicking of vertical route window expansion/contraction variable direction double chevron button 208. Note that the

direction of the double chevrons of vertical route window expansion/contraction variable direction double chevron button 208 is now pointing upward to designate that the configurable route window 204 can be contracted in an upward direction upon an additional click of vertical route window expansion/contraction variable direction double chevron button 208. Adaptive graphical map segment 202 has adapted its size and orientation to the newly reconfigured configurable route window 204. Display format parameter control buttons 206 remain unchanged in number and selection; however, the amount of lines of information displayed beneath each of the display format parameter control buttons 206 has increased.

Now referring to Figure 6, there is shown a display generally designated 600, which could be the result of either clicking the vertical route window expansion/contraction variable direction double chevron button 208 of Figure 3, or clicking the horizontal route window expansion/contraction variable direction double chevron button 212 of Figure 5. In either case, the configurable route window 204 has expanded to consume the entire viewing area of the display 600. Adaptive graphical map segment 202 (now not shown in Figure 6) has been adapted to be completely removed from view. Note that the directions of the double chevrons of vertical route window expansion/contraction variable direction double chevron button 208 and horizontal route window expansion/contraction variable direction double chevron button 212 both designate that the configurable route window 204 can only be contracted.

Throughout this description, the terms “buttons” and “clicking” have been used. They are selected because they are believed to readily convey the present invention; however, it should be understood that other visual marks, other than buttons, could be substituted, and actions other than clicking could be substituted as well. It is intended that the present invention and the claims below be read to include all variations of these concepts. The parameters shown in display format parameter control buttons 206 are merely exemplary of many other parameters and abbreviations for such parameters which could be used with the present invention. Configurable route window 204 is shown with its home position in the upper left-hand corner of the display. This is a matter of designer choice, which can be readily varied, depending upon the particular design criteria for a particular display.

The hardware and software to create the displays of the present invention are either well known in the art, or could be adapted, without undue experimentation, from well-known hardware and software, by persons having ordinary skill in the art, once they have carefully reviewed the description of the present invention included herein.

It is thought that the method and apparatus of the present invention will be understood from the foregoing description and that it will be apparent that various changes may be made in the form, construct steps and arrangement of the parts and steps thereof, without departing from the spirit and scope of the invention or sacrificing all of their material advantages. The form herein described is merely a preferred exemplary embodiment thereof.

CLAIMS

I claim:

1. An avionics FMS comprising:

a source of FMS route data;

a display coupled to said source of FMS route data for visually presenting said FMS route data; and,

a configurable route window, presented on said display.
2. An avionics display of claim 1 wherein said display presents an adaptive graphical map segment.
3. An avionics display of claim 2 wherein said configurable route window includes a vertical route window expansion/contraction button.
4. An avionics display of claim 3 wherein said vertical route window expansion/contraction button is a vertical route window expansion/contraction variable direction indicating button.

5. An avionics display of claim 4 wherein said vertical route window expansion/contraction variable direction indicating button is a vertical route window expansion/contraction variable direction double chevron button.

6. An avionics display of claim 3 wherein said configurable route window includes a horizontal route window expansion/contraction button.

7. An avionics display of claim 2 wherein said adaptive graphical map segment adapts to a sized characteristic of said configurable route window.

8. An avionics display of claim 7 wherein said configurable route window includes a plurality of display format parameter control buttons.

9. An avionics display of claim 8 wherein said plurality of display format parameter control buttons provide for pilot selectable content to be displayed in said configurable route window.

10. An avionics display of claim 1 wherein said configurable route window has a configurable size characteristic.

11. An avionics display of claim 1 wherein said configurable route window has a configurable content characteristic.

12. An avionics display of claim 10 wherein said configurable route window has a configurable content characteristic.

13. An apparatus of claim 9 wherein each of said plurality of display format parameter control buttons, includes a plurality of selectable regions thereon which are independently capable of being highlighted to indicate a selection status.

14. An avionics display comprising:

means for presenting a configurable window of route information relating to a predetermined route of an aircraft;

means for graphically presenting a map relating to information relating to said predetermined route of said aircraft; and

wherein said window is simultaneously presented with said map.

15. An avionics display of claim 14 wherein said means for presenting a configurable window includes a first means for size adjustment of said configurable window.

16. An avionics display of claim 15 wherein said means for presenting a configurable window includes a second means for size adjustment of said configurable window.

17. An avionics display of claim 14 wherein said means for presenting a configurable window includes a means for altering a content characteristic of said configurable window.

18. A method of adjusting a display of FMS information to a pilot comprising the steps of:

providing a display of FMS information in a tabular textual format in a section of an avionics display;

positioning a cursor over a predetermined position in said section and providing a selection input signal having a characteristic associated with said predetermined position; and

adjusting a characteristic of said section in response to said selection input signal.

19. A method of claim 18 wherein said step of adjusting a characteristic includes changing a size characteristic of said section.

20. A method of claim 18 wherein said step of adjusting a characteristic includes changing an information content characteristic of said section.

ABSTRACT

An apparatus and method for displaying FMS information to a pilot with a map segment and a configurable window of textual information which is configurable both in size and content.

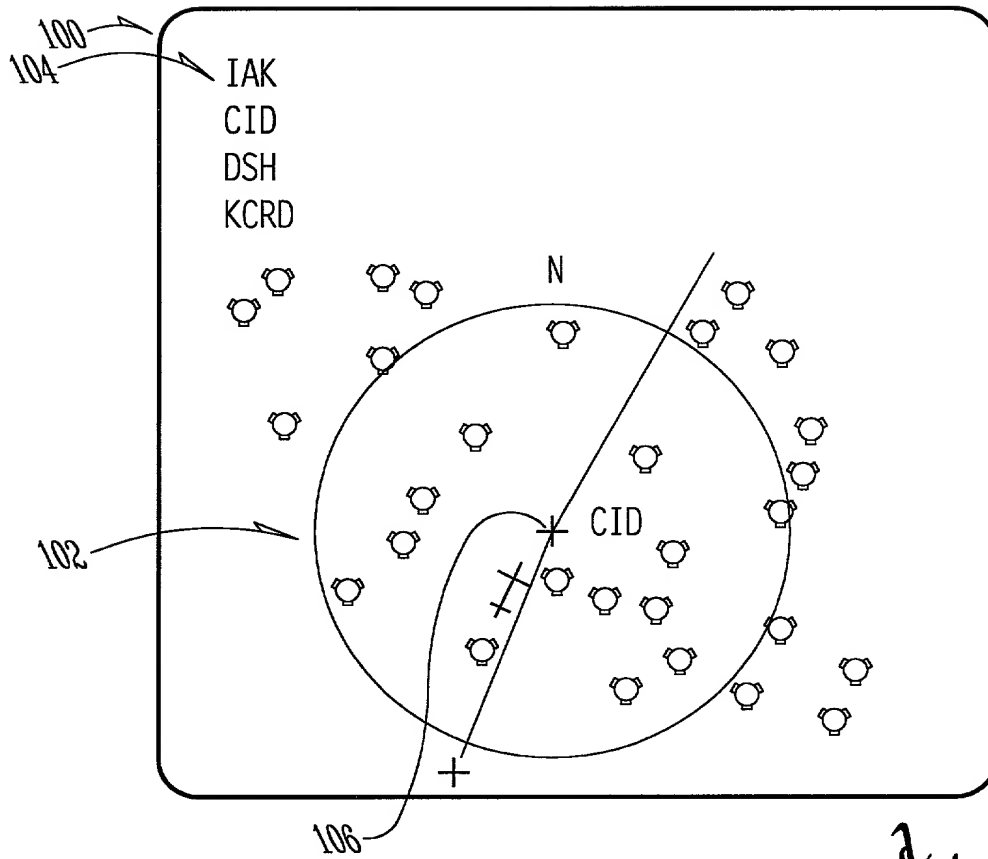
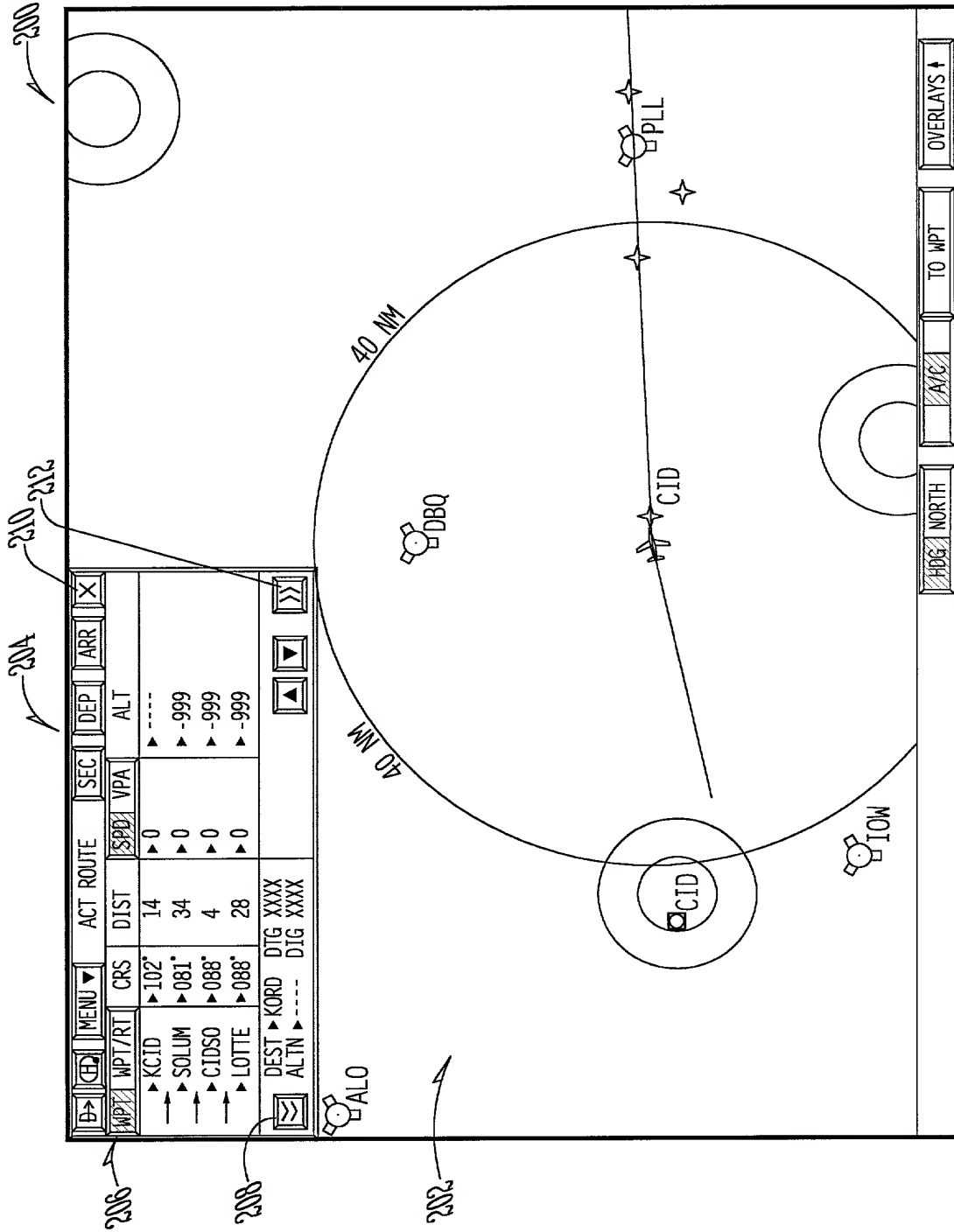


Fig. 1
(PRIOR ART)



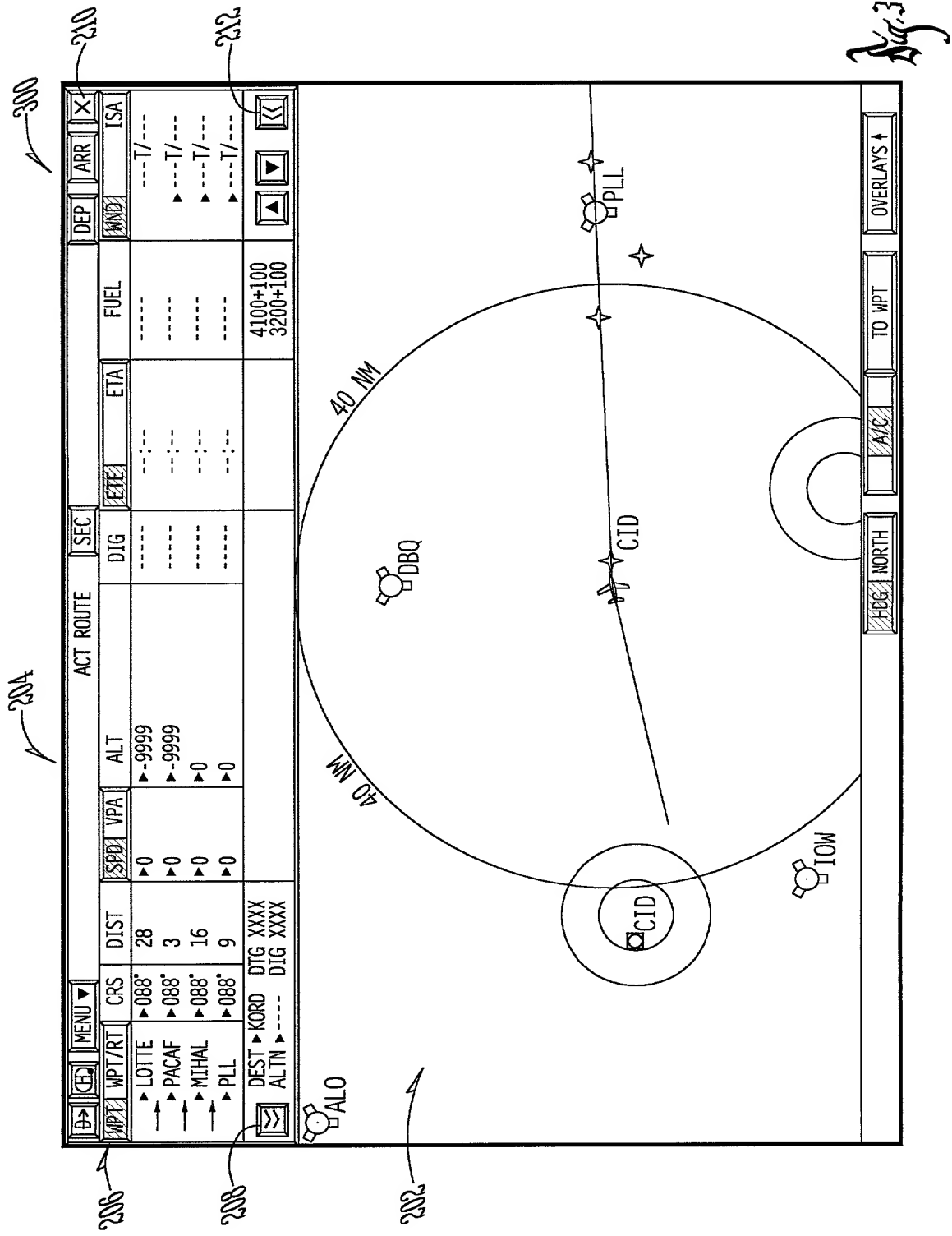


FIG. 4 is a schematic diagram of a flight information system (FIS) interface, showing a top-down view of a flight path and associated data.

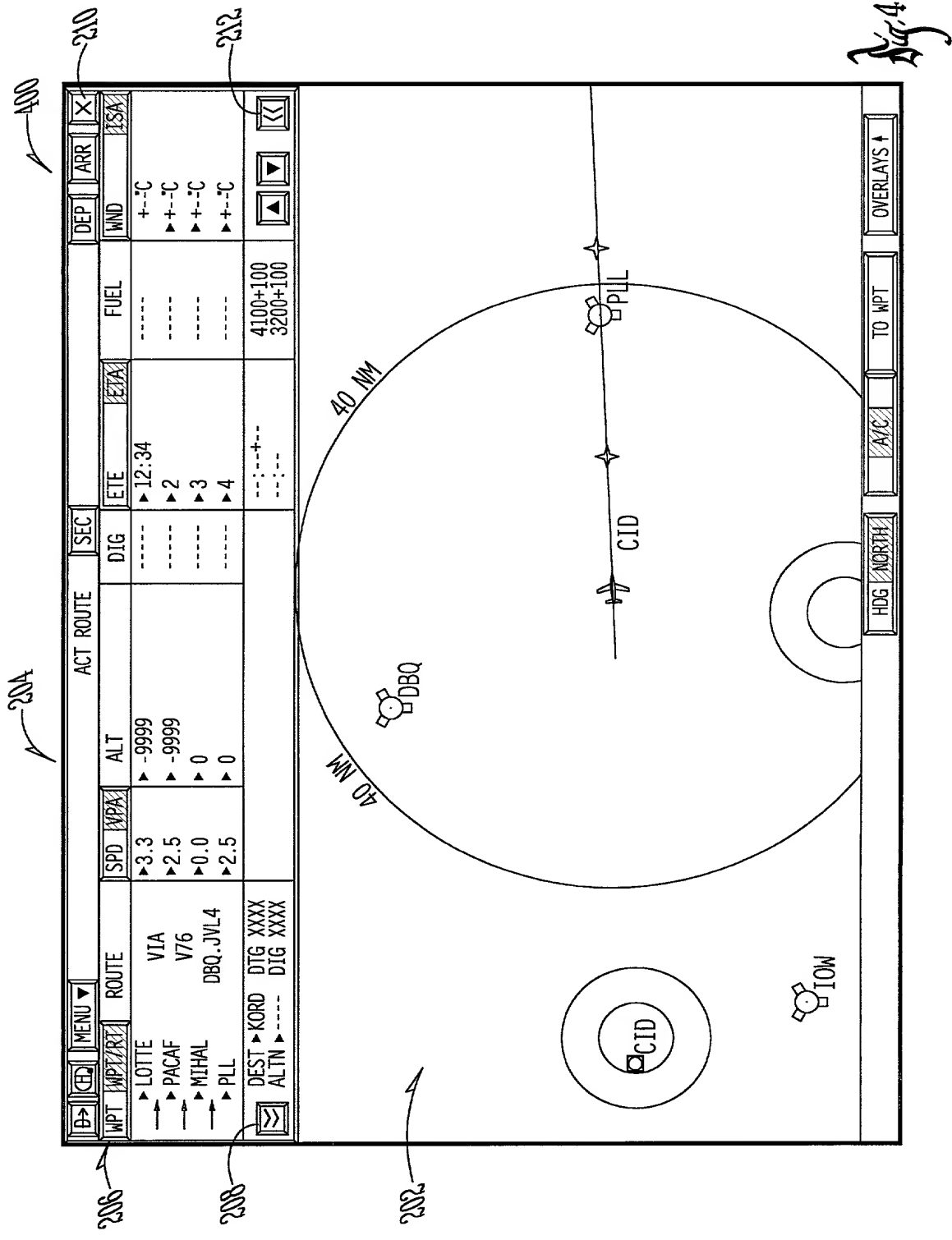
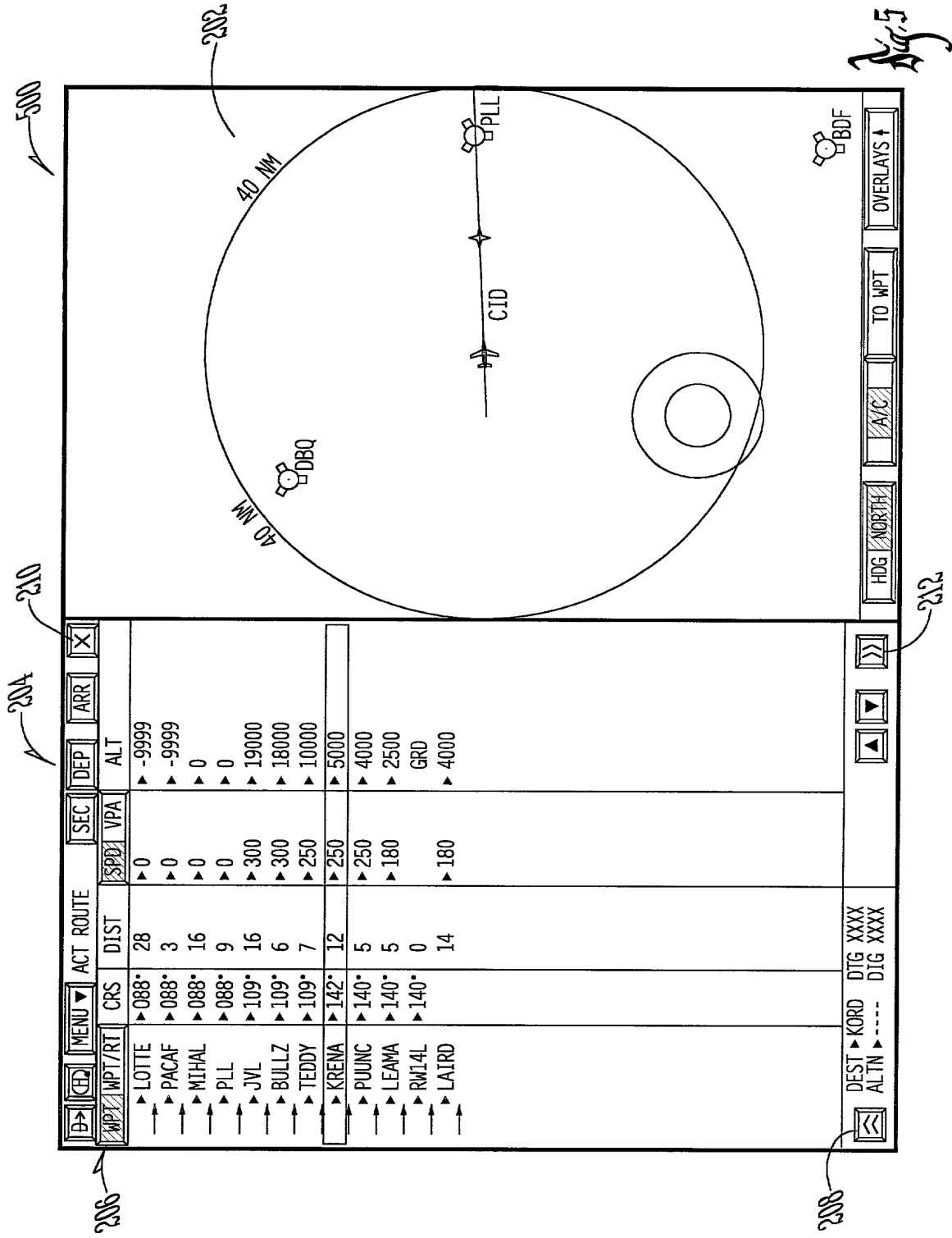


Fig. 4



204

600

210

ACT ROUTE										DEP	ARR	X
WPT	WPT/RT	CRS	DIST	SPD	VPA	ALT	DIG	ETE	ETA	FUEL	ISA	
↑	LOTE	088°	28	0		0						
↑	PACAF	088°	3	0		0						
↑	MTHAL	088°	16	0		0						
↑	PLL	088°	9	0		0						
↑	JVL	109°	16	300		19000	50.5	0:08		4280	010T/55	
↑	BULLZ	109°	6	300		18000	34.5	0:07		4240	010T/50	
↑	TEDDY	109°	7	250		10000	28.5	0:05		4205	020T/30	
↑	KREMA	142°	12	250		5000	21.5	0:03		4160	030T/25	
↑	PUUNC	140°	5	250		4000	9.5	0:04		4140	080T/14	
↑	LEAMA	140°	5	180		2500	4.7	0:03		4120	080T/14	
↑	RW14L	140°	0			GRD	0			4100	080T/14	
↑	LAIRD		14	180		4000				4050	030T/25	

212

212

206

208

DEST KORD DTG XXXX
ALTN DTG XXXX

DECLARATION FOR PATENT APPLICATION

Docket #99CR108/KE

As a below named inventor, I hereby declare that:
My residence, post office address and citizenship are as stated below next to my name.

I believe I am the original, first and sole inventor (if only one name is listed below) or an original, first and joint inventor (if plural names are listed below) of the subject matter which is claimed and for which a patent is sought on the invention entitled **Method And Apparatus For Interactively Displaying A Route Window For A Flight Management System**, the specification of which:

(check one) ☒ is attached hereto.
☐ was filed on _____ as
☐ Application Serial No. _____
and was amended on _____.

I hereby state that I have reviewed and understand the contents of the above identified specification, including the claims, as amended by any amendment referred to above.

I acknowledge the duty to disclose information which is material to the examination of this application in accordance with Title 37, Code of Federal Regulations, §1.56(a).

I hereby claim foreign priority benefits under Title 35, United States Code, §119 of any foreign applications(s) for patent or inventor's certificate listed below and have also identified below any foreign application for patent or inventor's certificate having a filing date before that of the application on which priority is claimed:

Prior Foreign Application(s) None

I hereby claim the benefit under Title 35, United States Code, §120 of any United States application(s) listed below and, insofar as the subject matter of each of the claims of this application is not disclosed in the prior United States application in the manner provided by the first paragraph of Title 35, United States Code, §112, I acknowledge the duty to disclose material information as defined in Title 37, Code of Federal Regulations, §1.56(a) which occurred between the filing date of the prior application and the national or PCT international filing date of this application:

(Application Serial Number)	(Filing Date)	(Status -patented, pending, abandoned)
(Application Serial Number)	(Filing Date)	(Status -patented, pending, abandoned)

I hereby appoint the following attorney(s) and/or agent(s) to prosecute this application and to transact all business in the Patent and Trademark Office connected therewith: Kyle Eppele, Reg. #34,155, James P. O'Shaughnessy, Reg. #27,667, John J. Horn, Reg. # 28,803, John M. Miller, Reg. # 38,560, John J. Deinken, Reg. # 28,406, and William R. Walbrun, Reg. # 37,464.

Address all telephone calls to Kyle Eppele at telephone no. (319) 295-8280

Address all communications to Rockwell Collins, Inc.

Attention: Kyle Eppele M/S 124-323
400 Collins Rd. NE
Cedar Rapids, Iowa 52498

I hereby declare that all statements made herein of my own knowledge are true and that all statements made on information and belief are believed to be true; and further that these statements were made with knowledge that willful false statements and the like so made are punishable by fine or imprisonment, or both, under Section 1001 of Title 18 of the United States Code and that such willful false statements may jeopardize the validity of the application or any patent issued thereon.

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